

# Engineering considerations in design of high temperature electronics for planetary probes

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# The Challenge for the Entry Phase in the Mission

## PRIMARY TECHNICAL CONSTRAINT:

1. Properties of binding materials – cannot exceed 260° C.
2. Minimize Thickness and weight of the TPS material – but meet 260° C constraint
3. Thus electronics, to be embedded in TPS, must be qualified to 260° C.

## TRADE-OFF PARAMETERS:

1. Thickness of material
2. Heating rates
3. Electronics technologies

## Illustrative Example ~ Properties for Orion:

### Lunar direct return conditions:

- 11 km/s atmospheric entry
- peak heat rate > 750 W/cm<sup>2</sup>

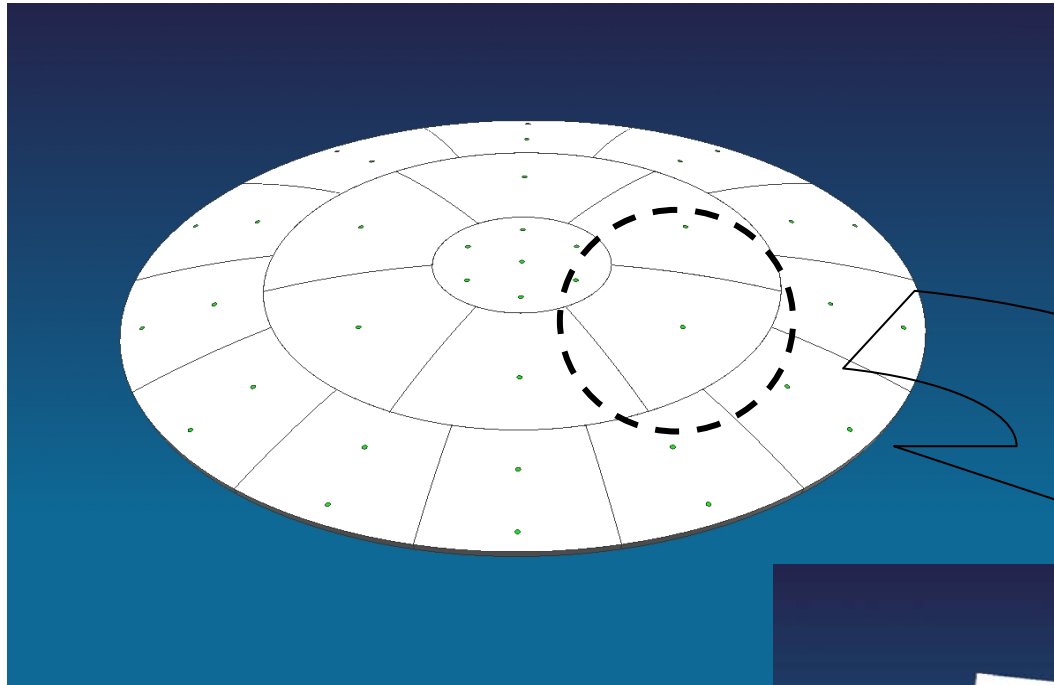
### LEO return conditions:

- 8 km/s atmospheric entry
- peak heat rate > 100 W/cm<sup>2</sup>

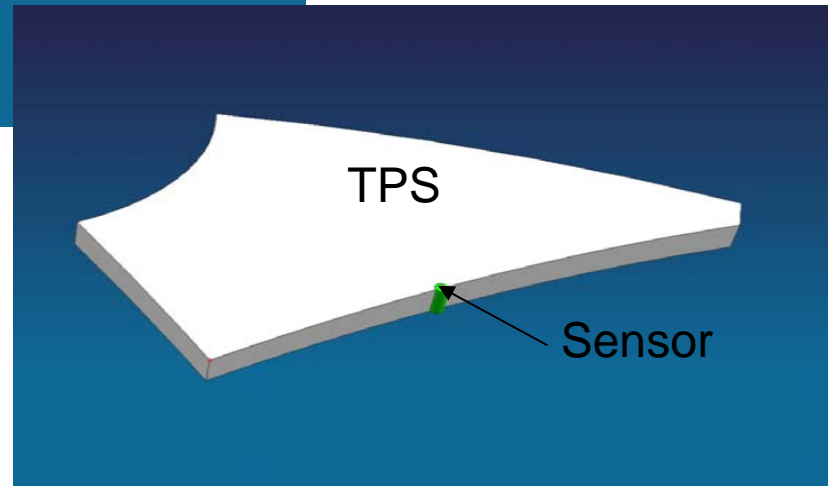


*Galileo Probe*

# Heat Shield Sensors



Multiple sensor modules would be embedded into an instrumented TPS material as implemented in Galileo, Orion, etc..



# CEV Heat Shield Sensor Architecture for Extreme Environments

## **Background (the technical problem to be solved):**

The TPS and heat shield will be generously instrumented during the atmospheric entry. Sensors could include, for example, 100 thermal couples, 50 recession sensors, 20 pressure sensors, and accelerometers. These sensors will require electronic circuits for interface and data collection. At locations behind the PICA TPS, the temperatures may reach, in worst-case conditions, up to 260° C = 500° F.

## **Background (A feasible technology):**

Silicon Germanium (SiGe) Electronics for Extreme Cryo Environments is being developed under NASA RHESE program managed by Dr. Andrew Keys, MSFC.

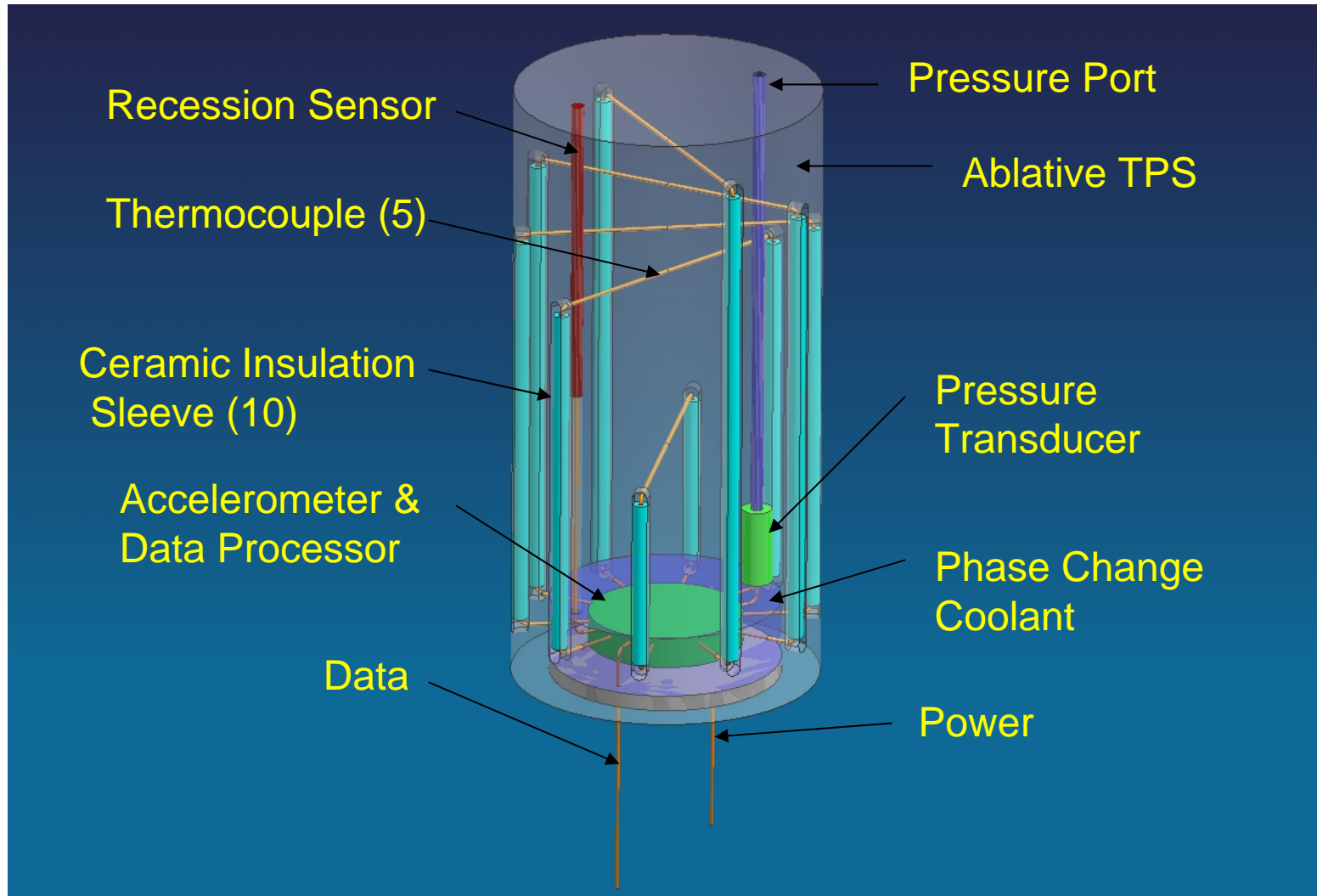
NASA program for SiGe Electronics for Extreme Environments performed by multi-disciplinary team (university, industry, NASA) led by Prof. John Cressler at Georgia Tech.

SiGe technology is currently being tested under Boeing IRAD to demonstrate its ability to operate at high temperatures. SiGe HBT devices also have intrinsic tolerance to radiation, as demonstrated in tests with multi-MRad total dose.

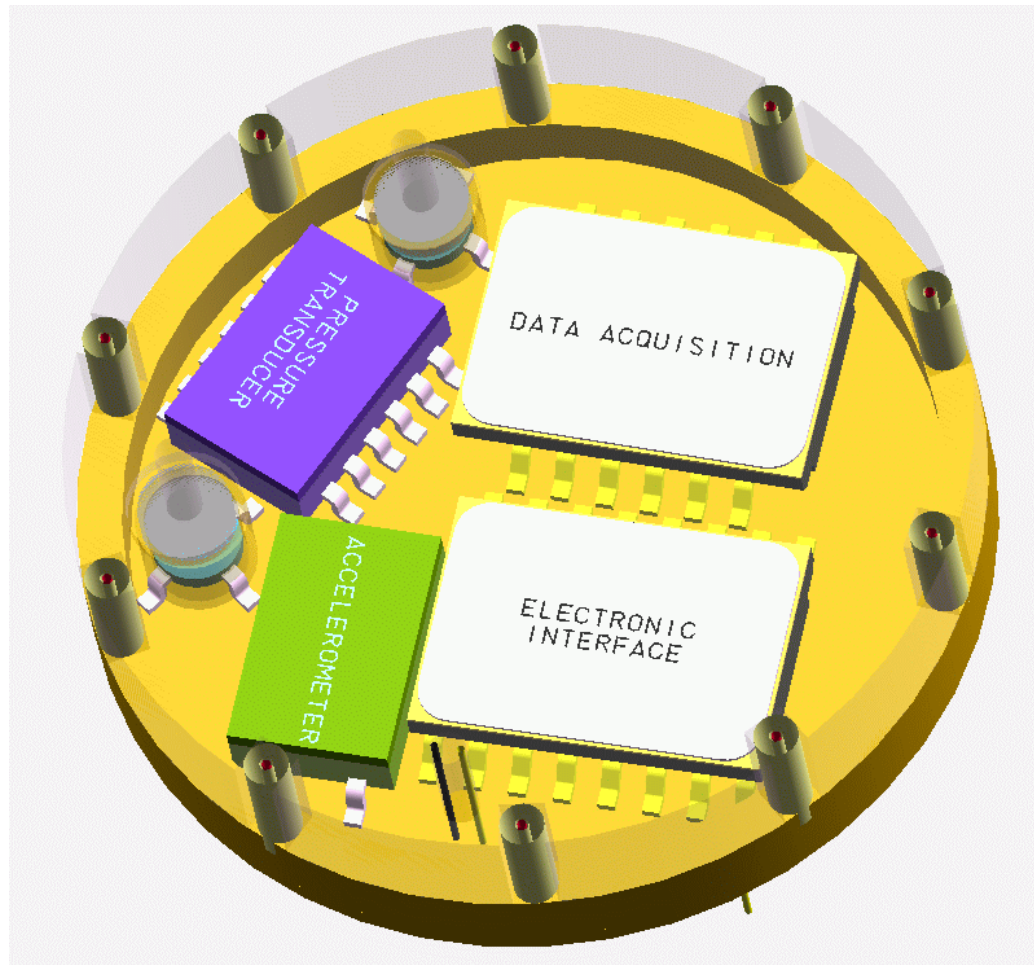
... by implementing integrated sensor measurement architecture using SiGe electronic modules, to operate without failure up to 260° C (500° F).



# Heat Shield Sensor Module



## SiGe Electronics Embedded in TPS Sensor Module



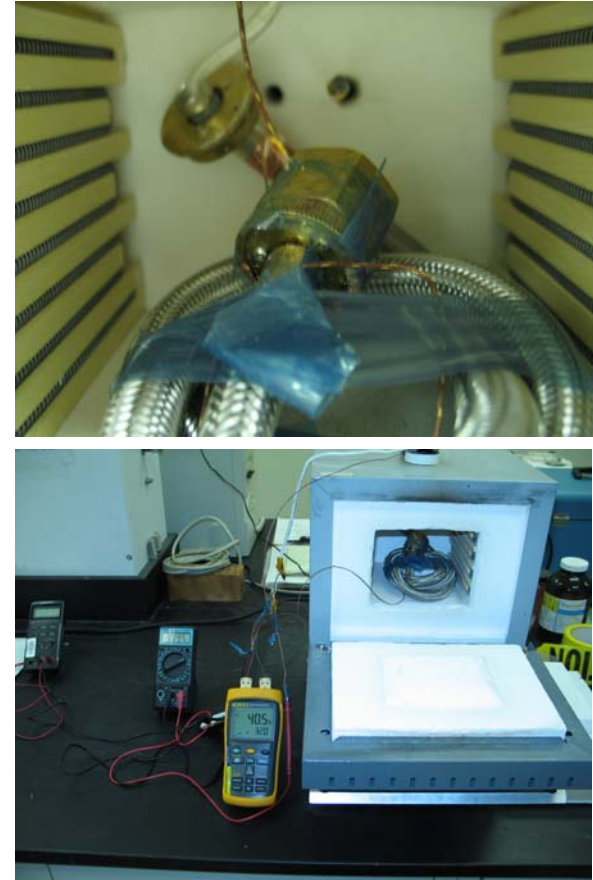
## **TECHNICAL APPROACH FOR VALIDATION OF INSTRUMENTED Heat Shield Sensor Architecture for Extreme Environments**

### **Technical Approach to Our Test Program:**

- 1. Test a representative SiGe circuit in oven tests, for validation of performance at high temperatures, to 260° C.**
- 2. Provide customized high temperature packaging for operation in the very high electronic noise inherent to an arc jet test site (this electric noise is not encountered behind a typical TPS structure during reentry). Test the same SiGe circuit, in a custom designed package for environment of high acoustic noise, behind the TPS panels.**
- 3. Design alternative (backup) package that incorporate phase-change materials, for demonstration of feasibility of additional margin.**
- 4. Test and demonstrate the representative SiGe circuit during arc jet tests. The SiGe electronics is powered and operational while embedded in the back of the PICA coupon during the arc jet test, in the environment where temperatures may reach 260° C (500° F).**



*Testing Si Ge at high temperature, using the oven in the lab*

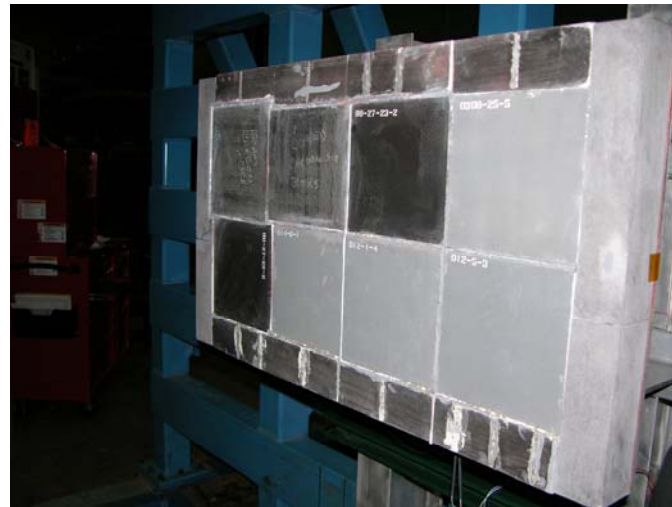


**Experimental setup in the Boeing Instrumental Analysis Lab**



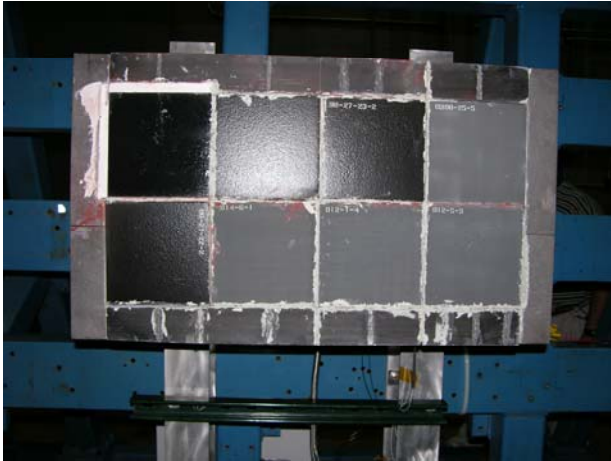
## *Testing of SiGe Electronics at high temperatures*

1. Tests up to 200° C, on unpackaged die, performed by Boeing together with the group of Prof John Cressler at Georgia Tech.  
Tests completed successfully, no evidence of functional degradation of electronics.
2. Acoustic and high-temp tests performed by Boeing at AFRL Dayton, on packaged SiGe circuit, housed within a module. Module located behind TPS (tiles) panel. TPS panel subjected to acoustic noise and front temperatures up to 2300° F.  
Tests completed successfully, no evidence of functional degradation of electronics.



TPS (tile) panel being tested at AFRL

## *Testing of SiGe Electronics at high temperatures at AFRL*



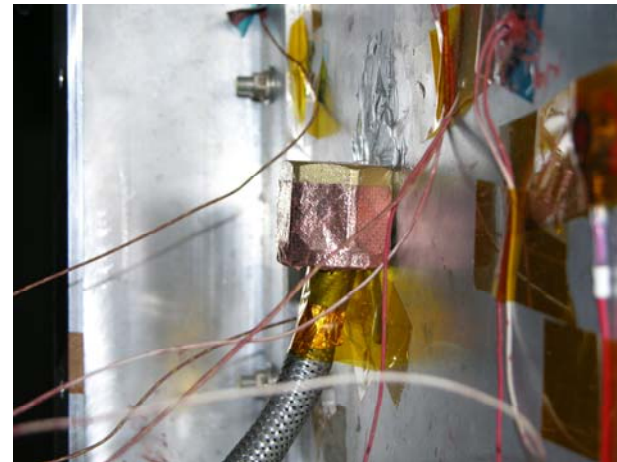
**TPS (tile) panel – front side in rack**



**Back side of panel, showing SiGe module and the Boeing test monitoring system.**

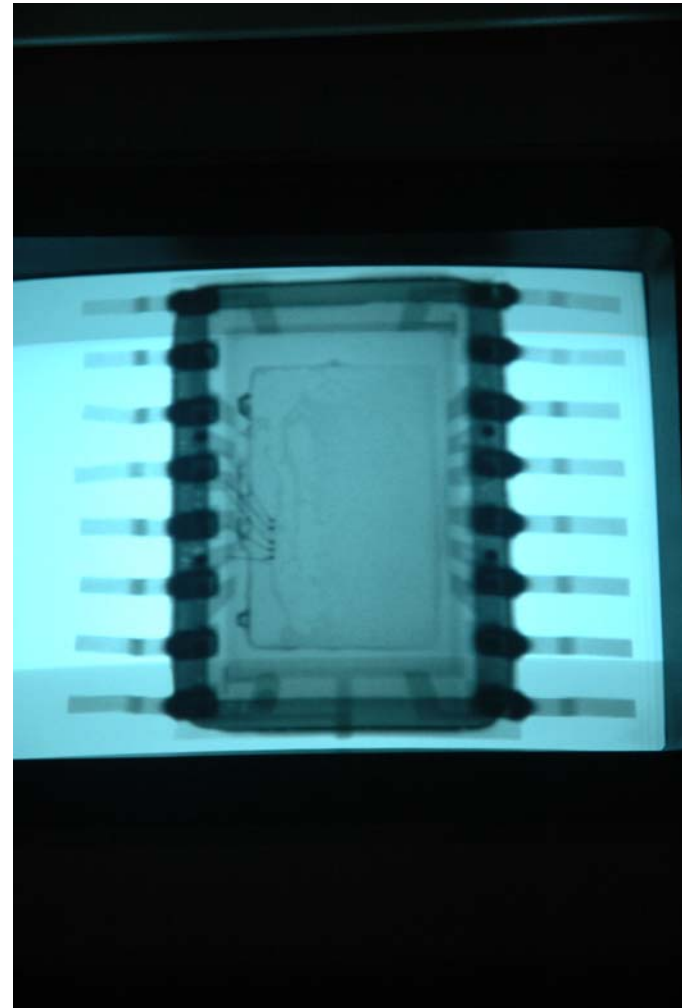


**The Boeing test monitoring system for the SiGe Electronic Module.**



**SiGe Electronic Module located on the back of the TPS (tile) panel.**

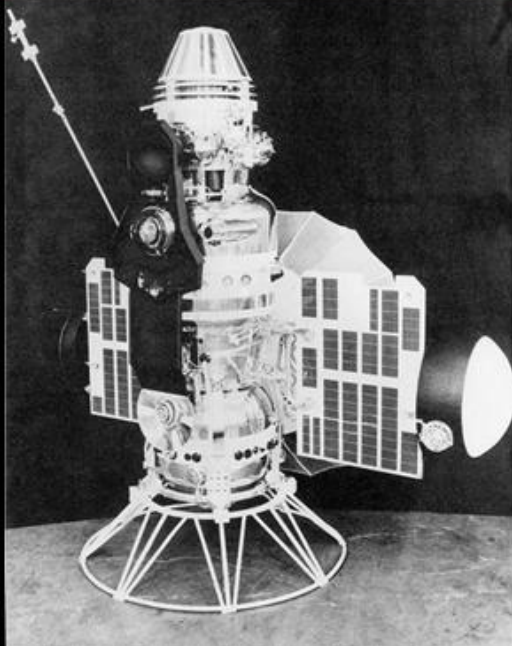
**X-Ray of SiGe module after  
Acoustics and Thermal testing.  
Testing and Oven Heat Soak at  
260°C for about 18 hours.  
Module was still functional. X-ray  
shows that there is no sign of  
degradation to the module.**



# Venus Environment: 480° C and 90 Bars Pressure at Surface

**Mission Life on  
Venus Surface:  
53-120 MINUTES**

**RUSSIAN SPACECRAFT  
VENERA-2 to VENERA-14**

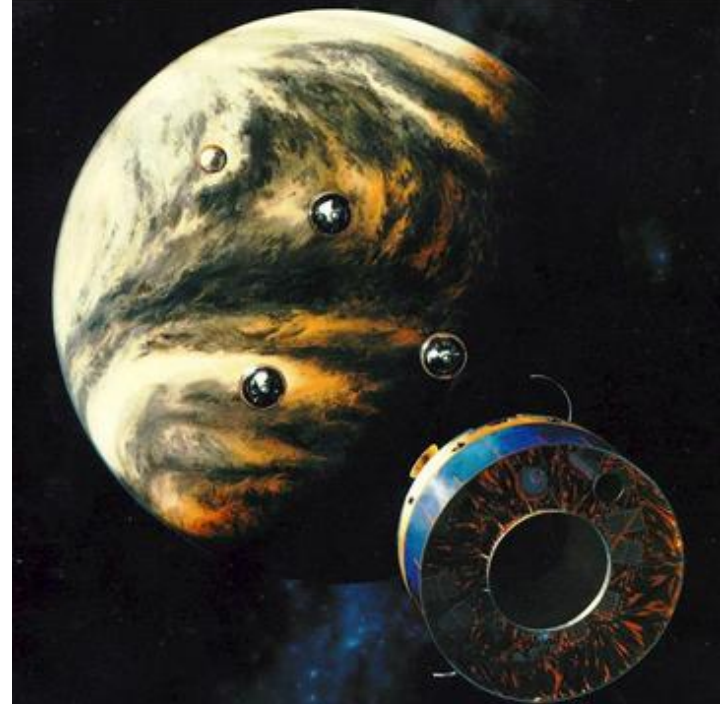


Venus Missions from the 1970s and 80s lasted under 2 hours on the Venus surface (electronics isolated in dewar from extreme heat).

NASA's "Venus Mobile Explorer" mission is required to have mission lifetime of 90 days on the Venus Surface.

Gallium Nitride (GaN) electronics could meet this 90 day mission requirement.

**PIONEER VENUS - MULTIPROBE**



**Mission Life on Venus Surface  
68 MINUTES**



## Illustrative concepts of future missions close to the Venus surface

This 90-day Venus Mission will be either a Lander/Rover or an Aerobot (pictured)



### **Aerobot Concept for Venus Mobile Explorer**

This long-lived in-situ Flagship class mission would provide aerial mobility close to the surface (approximately 10 kilometers above the surface) using metallic bellows to tolerate the extreme environment of Venus, where the temperature reaches 460 degrees Celsius, and the pressure is up to 90 bars, and the super critical carbon dioxide atmosphere is highly corrosive.

# High Temperature Electronics Technologies

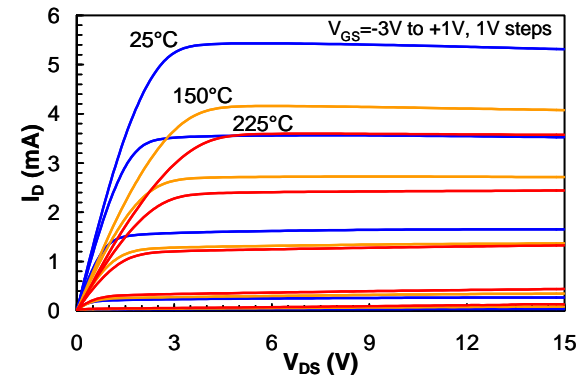
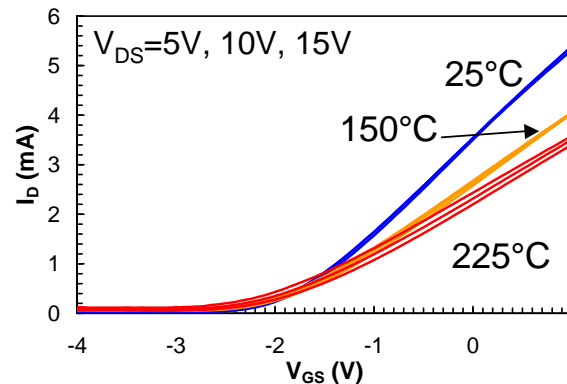
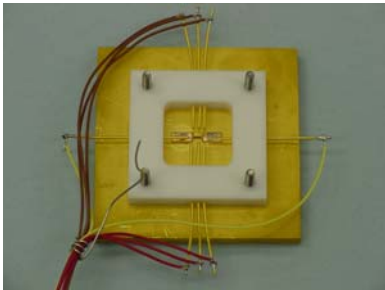
## Candidate Technologies for High Temperature Operation (and Maturity Level)

Technology	Temperature	Radiation tolerance	Integration level	Maturity
Silicon ~1.1eV	~125°C	Needs hardening	High	Very mature
SOI / SiGe ~1.1eV	~225°C (goal: 260°C )	Hard	Med-high	Mature / Maturing
SiC ~3.2eV (4H)	>300°C (goal: 600°C)	Hard	Discrete power	Emerging
GaN ~3.4eV	>350°C (goal: 500°C )	Very hard	Medium	Emerging

GaN ICs can provide solutions in integrated power and high temperature, high voltage, high radiation environments

# Characteristic behavior of GaN technology over temperature

***10 $\mu$ m/1 $\mu$ m GaN HFET Fabricated at HRL:  
Robust operation demonstrated to 225°C***

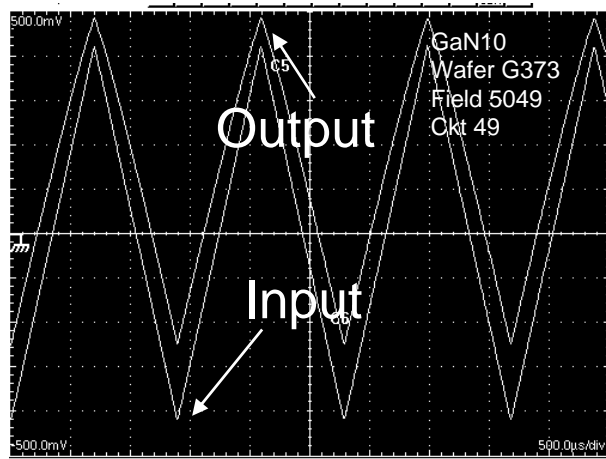


## **Status in 2007-2008:**

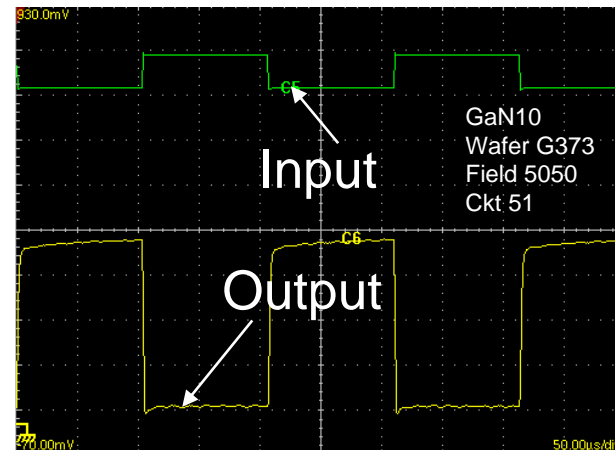
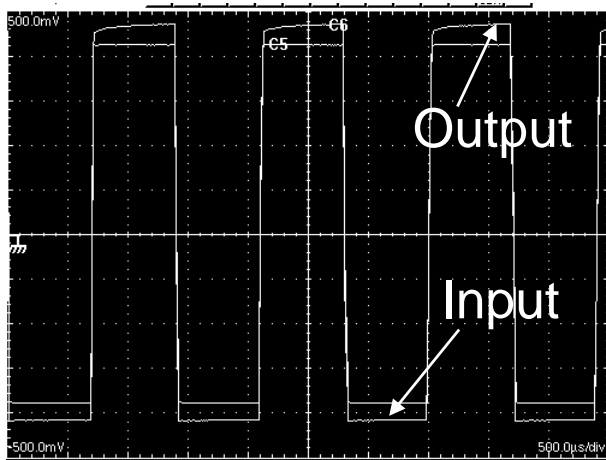
- Screened Integrated Circuits to 225°C operating temperatures
- HRL test fixture has capability to 300°C, successful exploration at 375°C
- Test capability at Boeing Huntington Beach, up to 500°C, in nitrogen flow.
- Nominal -1.8V threshold voltage shows no sign of short channel effects
- Unit transconductance scales well with device size.
- Both on-resistance and output conductance also scale well with device size



# Illustrative examples of GaN Integrated Circuits



**GaN Op-amp in unity gain and inverting gain configurations, driving 50Ω load (100mV/div)**

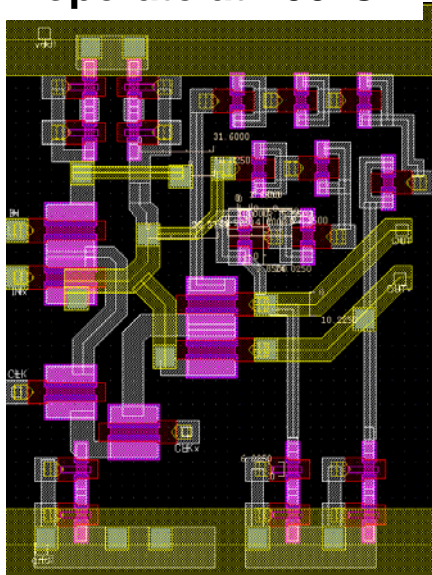


A General-purpose GaN Op-Amp Suitable for Operation in Harsh Environments

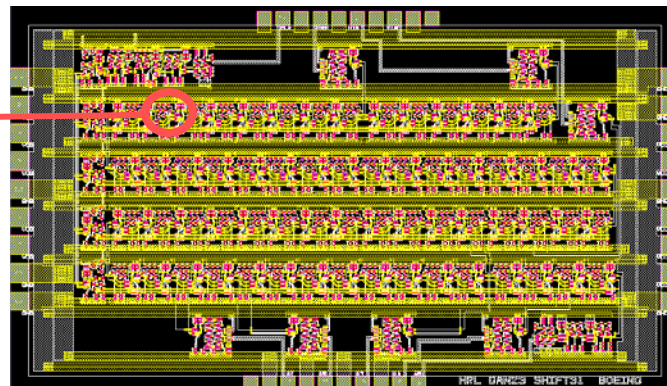
## Illustrative examples of GaN Integrated Circuits

- Consists of 3 implementations of shift register
  - 4-bits (418 transistors) in a pad frame picked for packaging
  - 15-bits (1084) and 31bits (1960) with external configuration as PRBS as an option
- Copies of two different verified designs of OP-Amps from GaN18
- Provision for backside plated metal for high-temperature packages

Latch designed to operate at 200°C



31 bit shift register



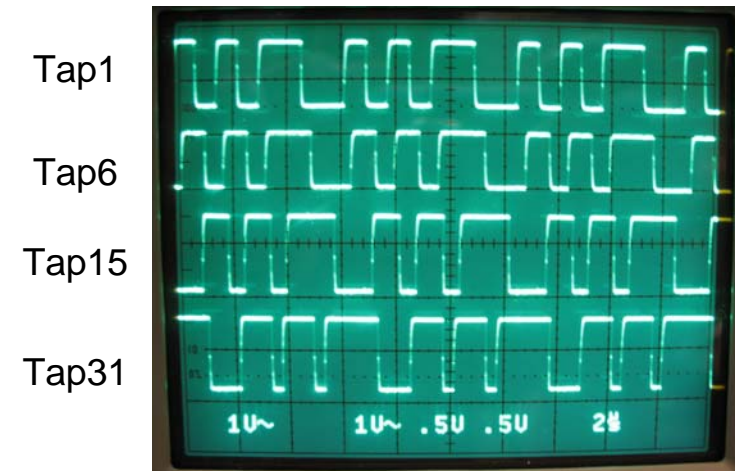
GaN23



4 bit shift register

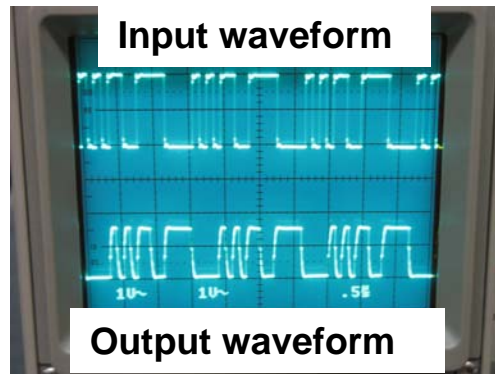
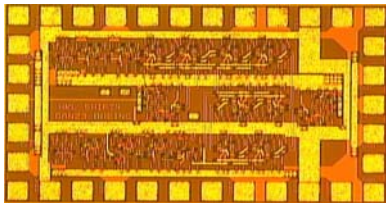
## Illustrative examples of GaN Integrated Circuits

- Basic functionality screen for Shift4 done
  - Operation verified to 10MHz
  - Yield on G536 wafer is ~80%,
  - G490 (had epi-material defects) yield ~50%
- Shift31 functionality verified on G536
- Ongoing testing, to understand yield



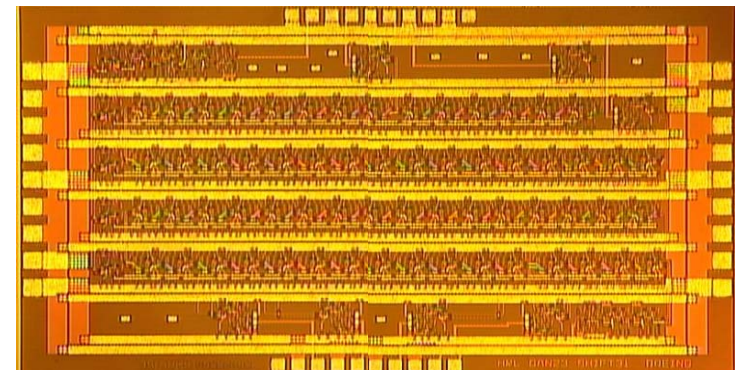
Shift31 test result

Micrograph of 4-bit shift-register



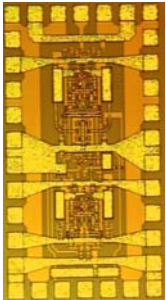
Shift4 test result

Micrograph of 31-bit shift-register

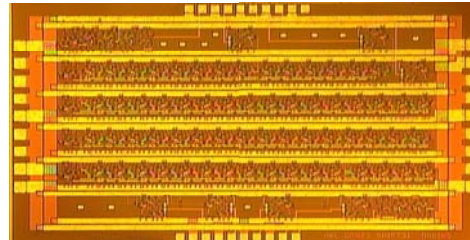




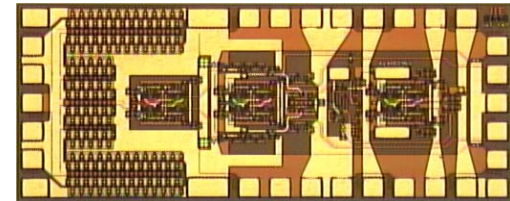
## Illustrative examples of GaN Integrated Circuits (cont.)



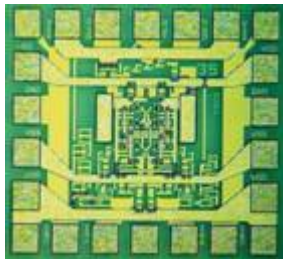
**GaN Op-amps**



**GaN shift register**

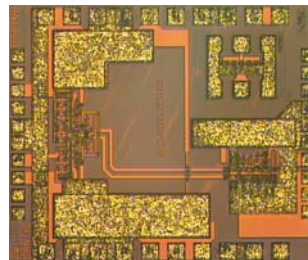


**GaN Low-Dropout  
Voltage Regulator**



**GaN Voltage  
Reference IC**

**GaN switch  
driver IC**



Demonstrating mixed-signal  
and power system  
capabilities in GaN IC  
technology